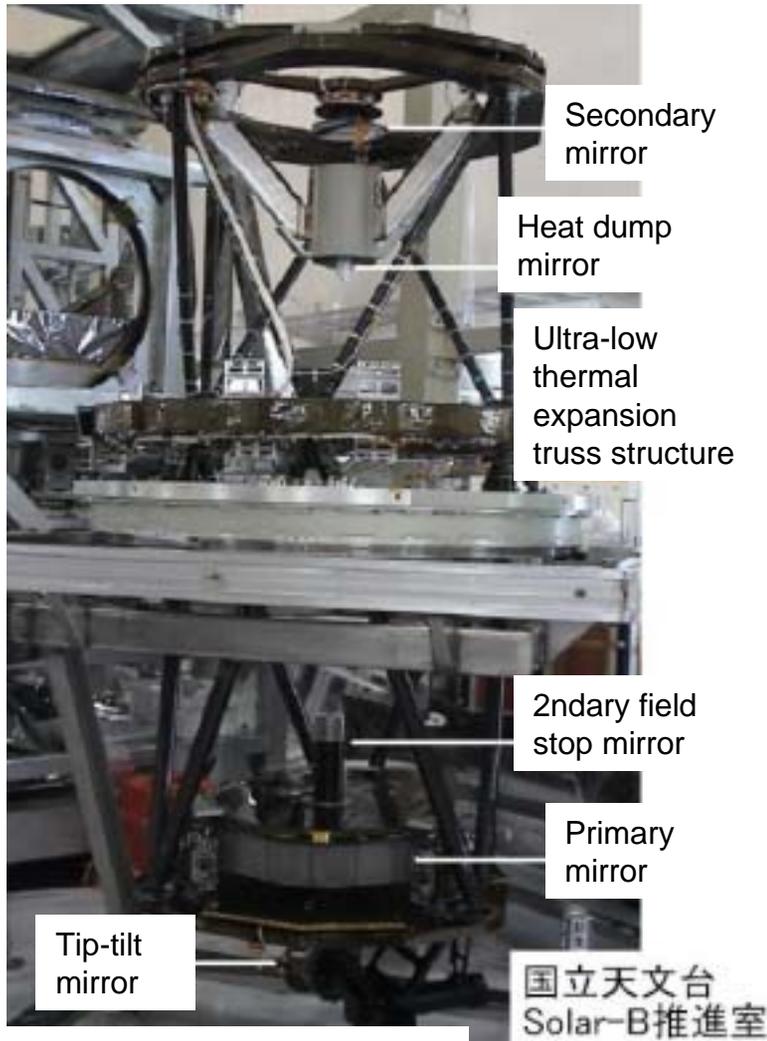


Optical Performance of OTA: Wavefront Error Measurements

Y. Suematsu (NAOJ) and OTA team

OTA Structure

Aplanatic Gregorian with heat dump mirror and collimator lens unit



Inside view of OTA

SOT#1 / Extended, 1 / Apr. 2006

- Low-expansion CFRP honeycomb panels and pipes (0.1ppm/K) are used for mirror cell, center section, ring plate, top ring, upper truss and lower truss.
- High thermal conductive CFRP is used to alleviate hot temperature spot caused by oblique incidence of solar light.
- Panels and pipes are adhesively bonded instead of connection by flange for weight saving and to avoid differential CTE.

CFRP= Carbon Fiber Reinforced Plastics

60cm flat mirror

rotation

tilt

OTA upward configuration (+1G)
for interferometer measurements

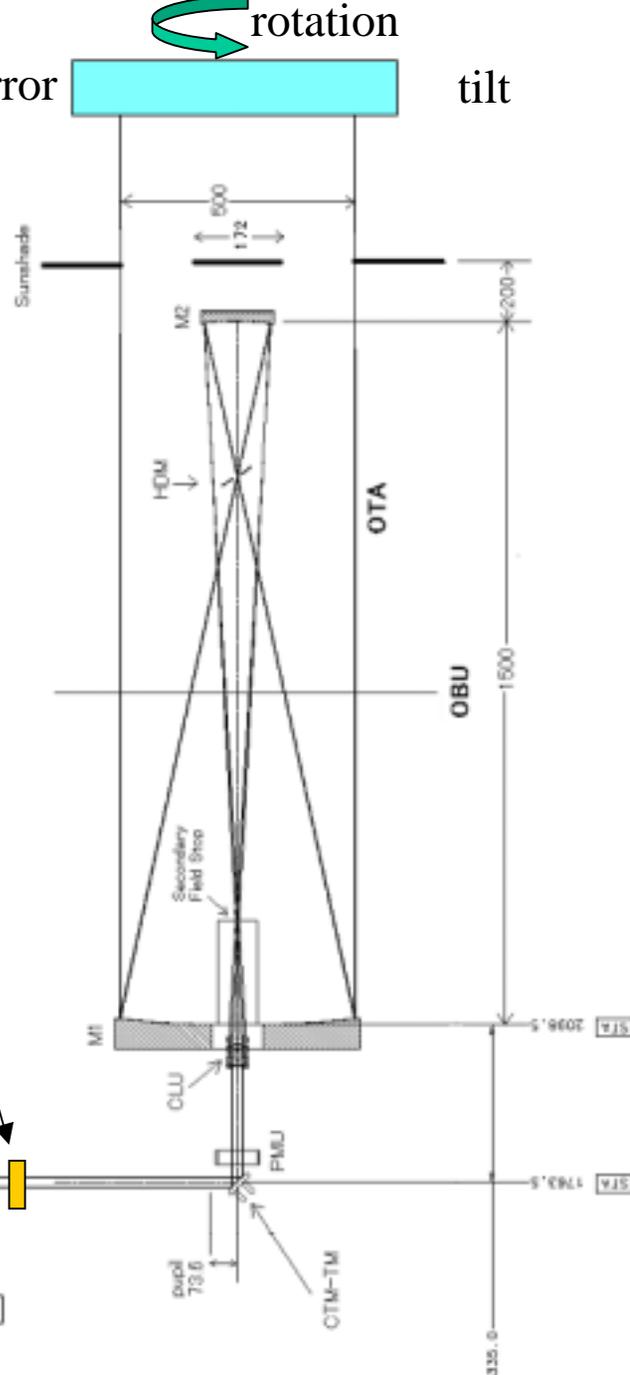
WFE in off the center field (~130") can be
measured by tilting the 60 cm folding flat

Measurements for OTA upside-down
configuration (-1G) are also performed

ND (pericle)
or polarizer

interferometer

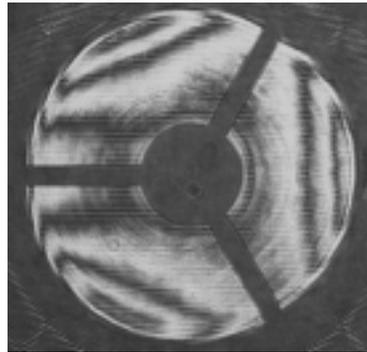
6-degree-of-freedom
active stage



FM-OTA Interferometer Measurements

Method

- usage of high-speed interferometer and phase map construction by spatial heterodyning method tolerant of vibration and seeing effect
- selection of good fringe map and average over 30-40 phase maps (FLIP bitmap mode)
- Consideration of air-flow system to make air temperature in optical pass uniform
- Zero-G optical performance is evaluated by an average of OTA upward and upside-down WFE measurements
- Usage of polarizer or ND in front of interferometer or off-axis measurement for rejection of ghost fringe from PMU
- Calibration of polarizer and ND WFE
- Calibration of 60cm folding flat mirror; non-axially symmetric component of WFE was removed by an average of 90 degs step rotation

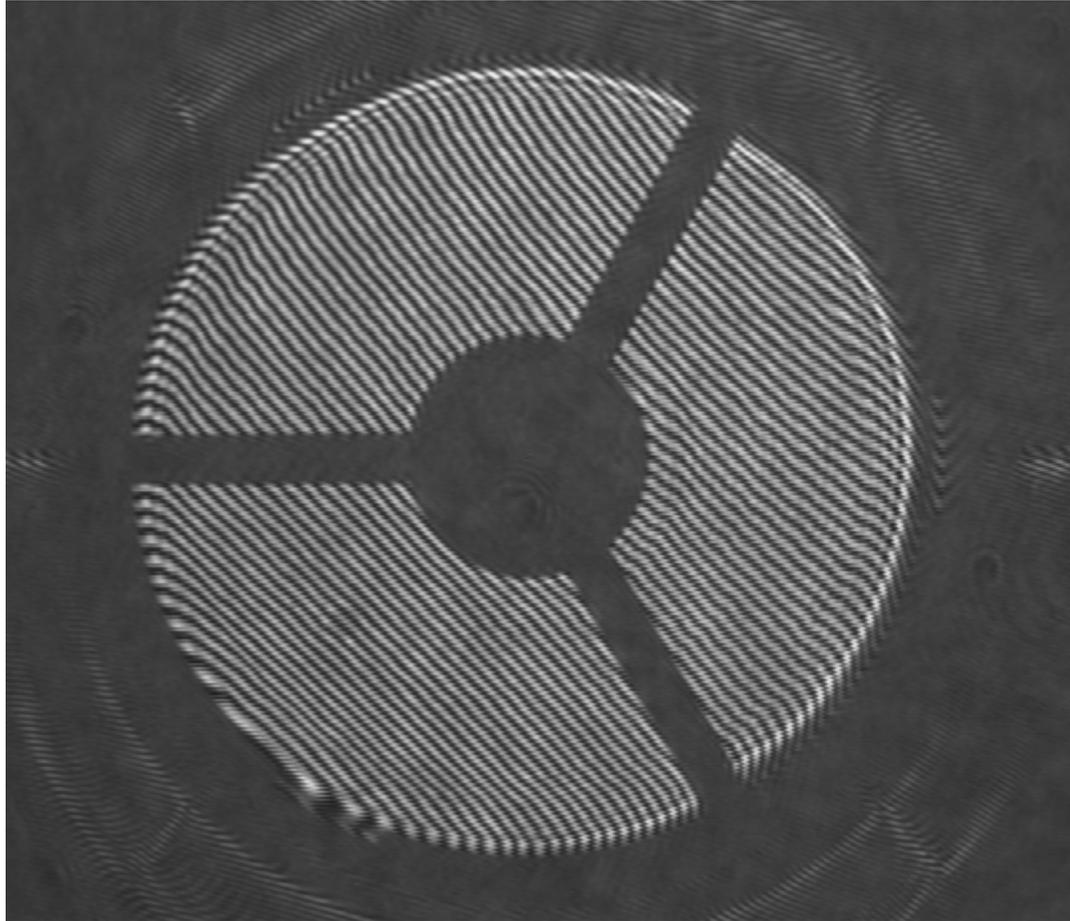


Null fringe map. Tri-angular astigmatism due to gravity is prominent.

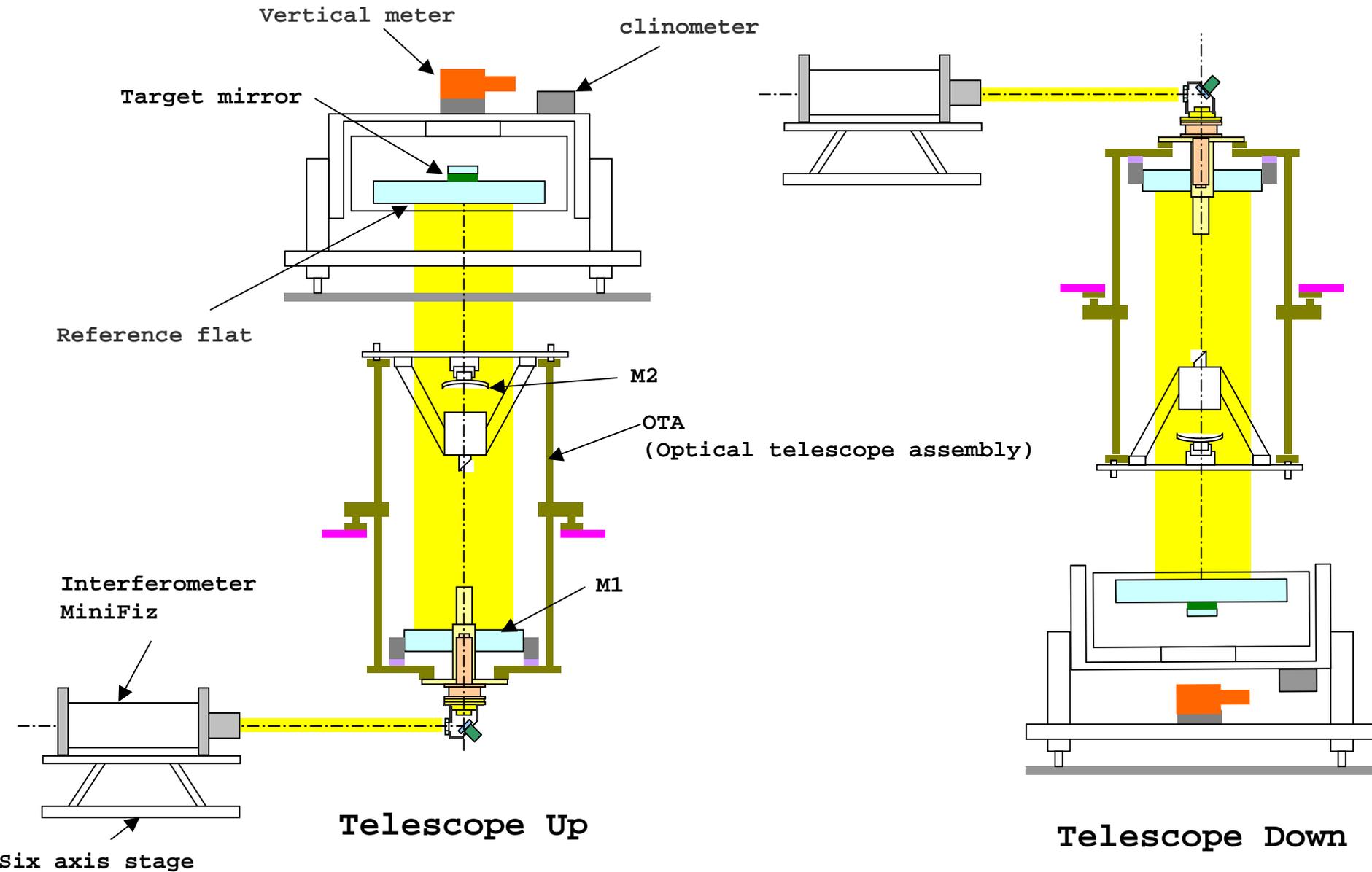


Measurements in upward (+1G) configuration

Sample of fringe for phase map derivation



OTA 0-G Test Configuration



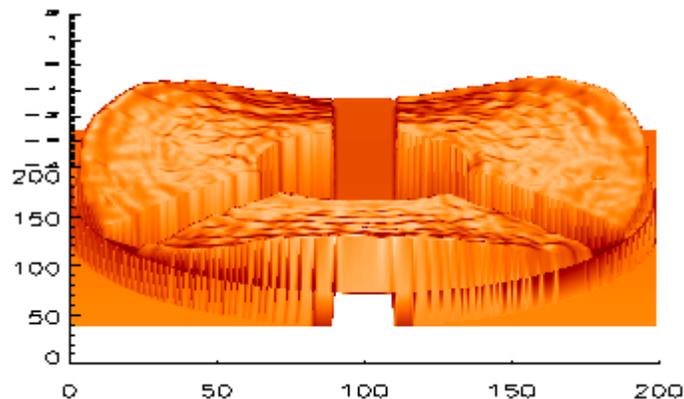
OTA Optical Test Configuration

Gravity deforms OTA primary mirror and CFRP truss!

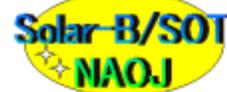


uppl-ref
26/04/05, 11:25:46
(unit: λ)
max = 1.2293
min = -2.3812
mean = -0.0132
PV = 3.6105
RMS = 0.7434

Zernike coefficient:
A00= 0.0619 A11= 0.0759 B11= 0.0334
A20= -1.4300 A22= 0.0397 B22= 0.0110
A31= 0.0545 B31= -0.0017 A40= 0.1122
A33= -0.0155 B33= 1.1246 A42= -0.0291
B42= 0.0144 A51= -0.0117 B51= 0.0351
A60= 0.0696

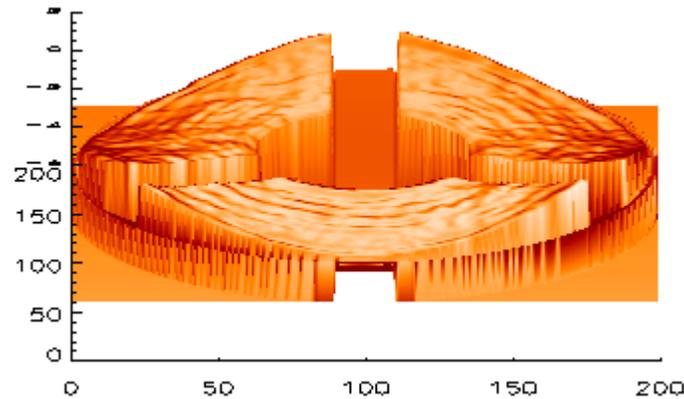
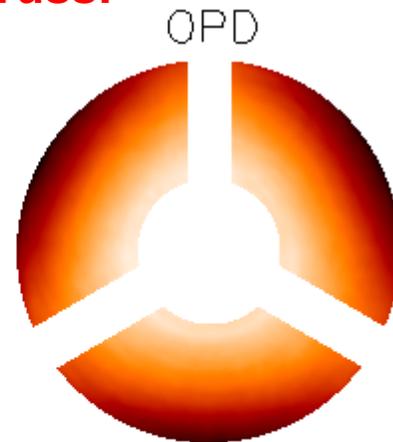


WFE for OTA upward configuration (+1G)
(average of 4 data from 90 degs step
rotation of 60 cm flat after calibration)



sh_dwpl-ref
00/00/00, 00:00:00
(unit: λ)
max = 1.8159
min = -2.7468
mean = 0.0013
PV = 4.5627
RMS = 1.0685

Zernike coefficient:
A00= 0.2411 A11= -0.0865 B11= 0.0534
A20= -2.0624 A22= -0.0126 B22= -0.0262
A31= 0.0780 B31= -0.0617 A40= -0.0478
A33= -0.0412 B33= -1.2193 A42= -0.0262
B42= 0.0122 A51= 0.0028 B51= 0.0241
A60= 0.0712



WFE for OTA downward configuration (-1G)
(average of 4 data from 90 degs step
rotation of 60 cm flat after calibration)

Optical Performance of OTA in 0-G

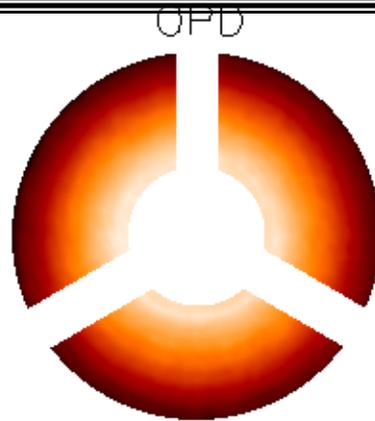
Solar-B/SOT
NAOJ

ALL
26/04/05, 11:31:49
(unit: λ)
max = 1.4946
min = -1.5728
mean = 0.0088
PV = 3.0674
RMS = 0.8104

Zernike coefficient:

A00 = 0.1591	A11 = -0.0061	B11 = 0.0420
A20 = -1.7427	A22 = 0.0138	B22 = -0.0089
A31 = 0.0687	B31 = -0.0368	A40 = 0.0380
A33 = -0.0285	B33 = -0.0491	A42 = -0.0269
B42 = 0.0120	A51 = -0.0020	B51 = 0.0256
A60 = 0.0755		

Average WFE of upward and
downward OTA

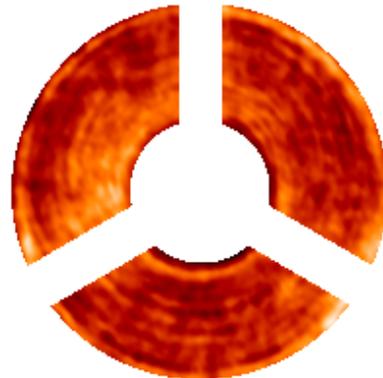


Solar-B/SOT
NAOJ

ALL
26/04/05, 11:31:49
(unit: λ)
max = 0.2336
min = -0.2064
mean = -0.0000
PV = 0.4401
RMS = 0.0572

subtracted

tilt and focus term removed.



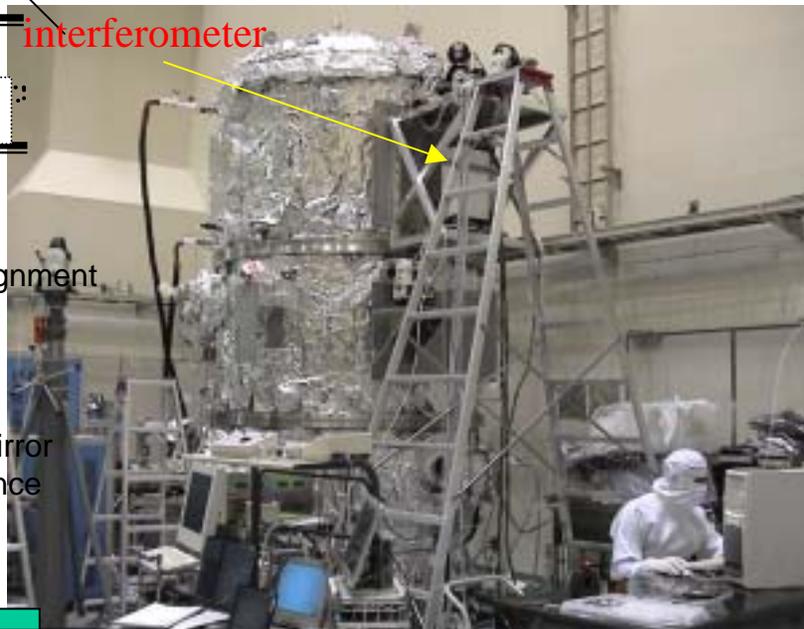
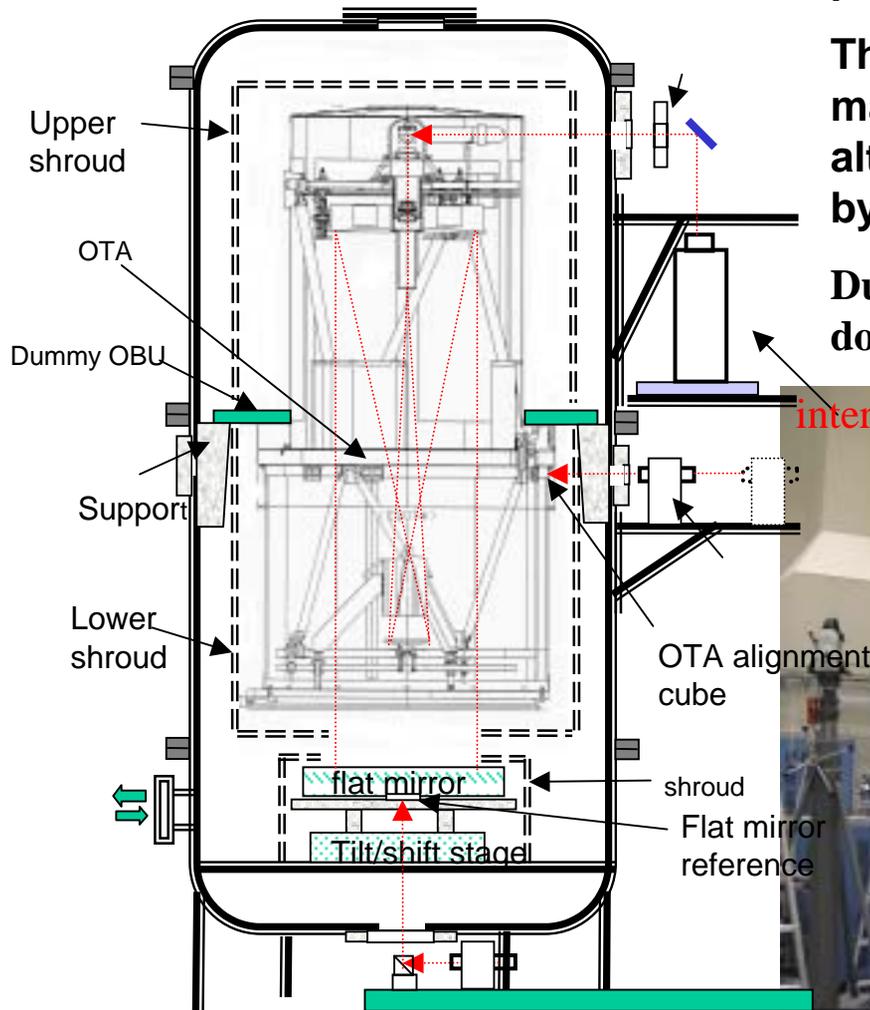
OTA WFE map for zero-gravity condition. This was obtained by averaging upward and downward OTA WFE. Then, rms WFE gives 0.0288 waves (**18.1nm, /35**) when focus error which can be corrected on-orbit are removed. This indicates that the rms WFE in the shortest observable wavelength 388 nm is 0.0469 waves and OTA WFE satisfies diffraction-limited condition (less than 0.071 waves, /14) in all the SOT observable wavelengths.

OTA Opto-Thermal Test for On-Orbit Performance

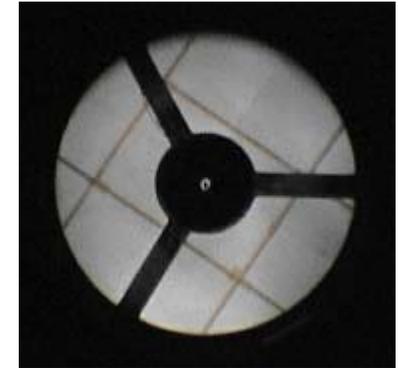
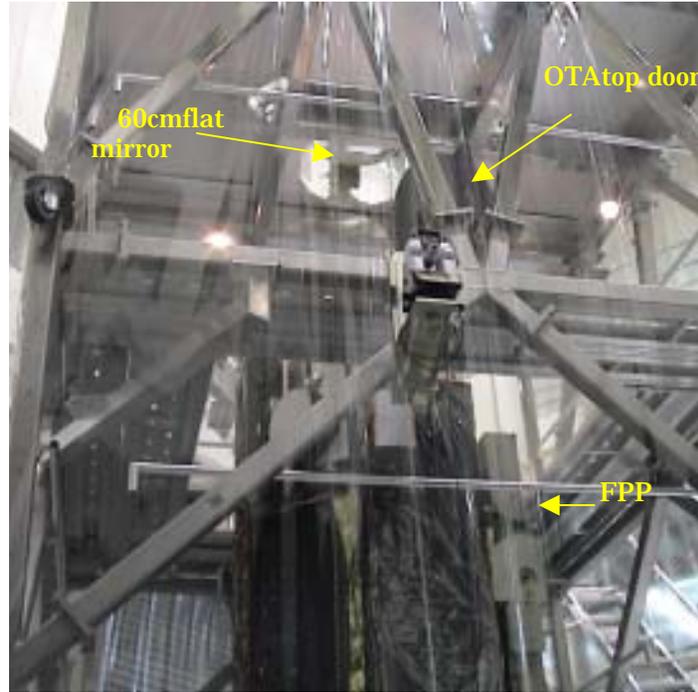
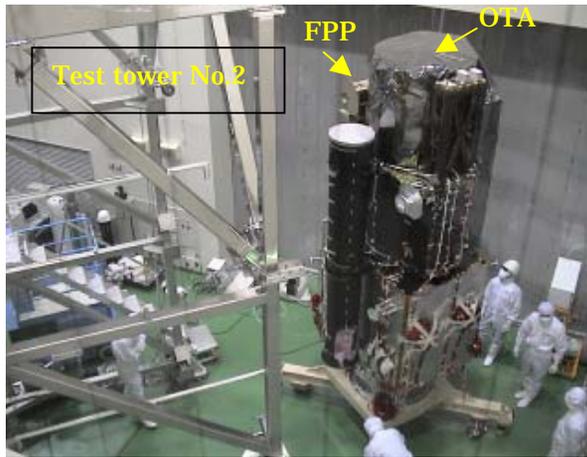
OTA WFE's were measured with an interferometer for cases that OTA had temperature distribution (- 30 ~ + 40) predicted for on-orbit condition.

The test demonstrated that OTA on orbit maintains diffraction limited performance although it was slightly deformed (14.7 nm rms) by the temperature change.

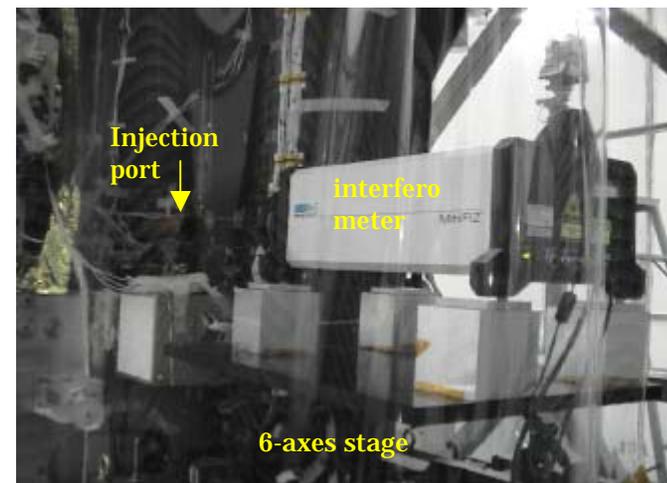
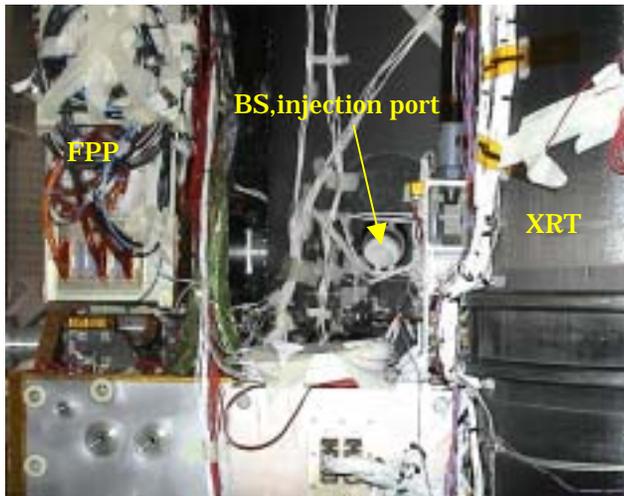
Duty cycle of operational heaters for M2 and CLU does not affect the optical performance of OTA.



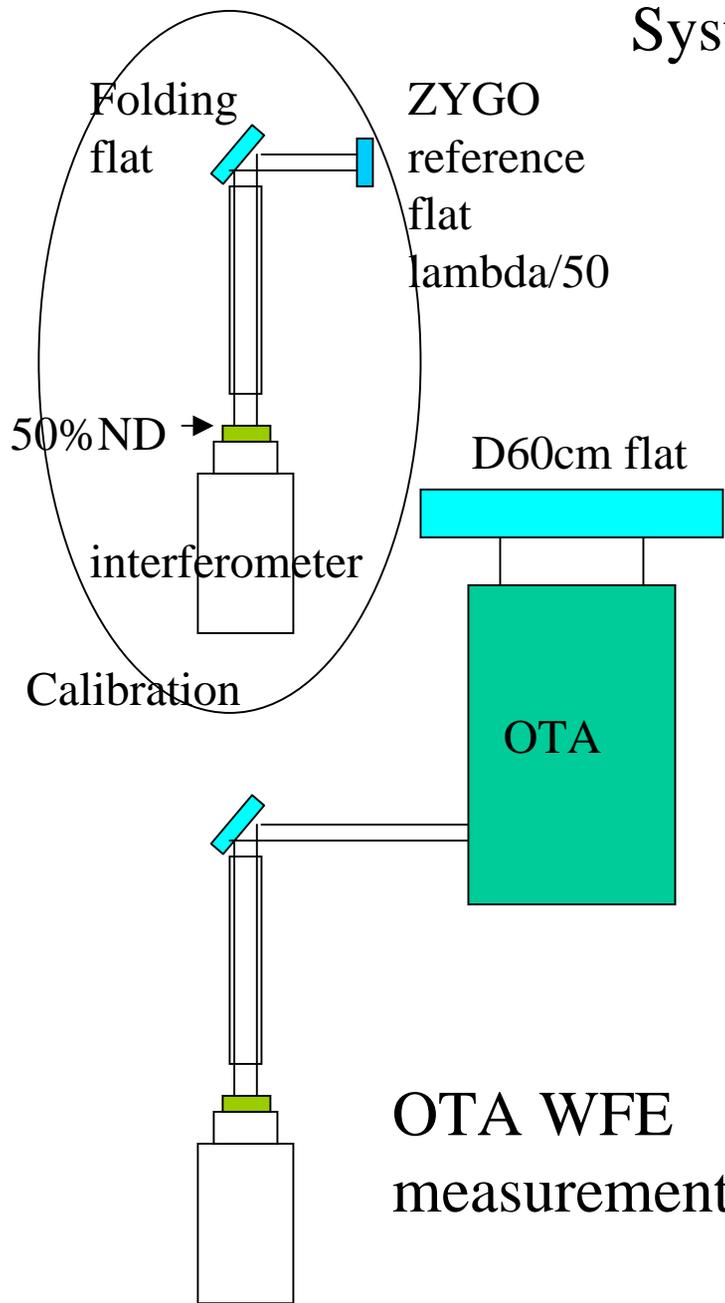
System-level optical test



OTA Pupil image

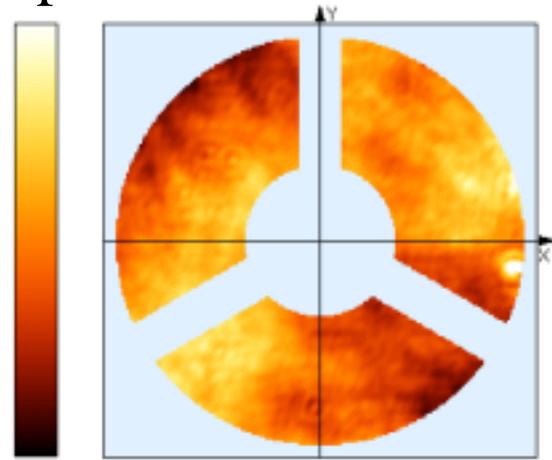


System-level OTA optical test



REOSC
SAGEM
 WaRPP v 3.00

av_refND
 Date : 23/08/05
 Time : 19:21:48
 Wavefront
 L = 632.80 nm
 R = 100.000 nm
 Resol. : 200x200
 Scale Lin. 1
 -59.161 nm to
 59.161 nm
 20900 points
 Min = -52.443 nm
 Max = 163.166 nm
 Avg = 0.000 nm
 P-V = 215.609 nm
 RMS = 19.720 nm



Calibration of 5 cm folding flat WFE

// Tilt	-0.004972	-0.002077
// Focus	-0.005948	
// Astig	0.017461	0.053928
// Coma	0.026772	-0.010023
// Sphere 3	-0.003328	
// Astig Tri	0.011442	0.001463
// Astig 5	-0.008874	-0.008480
// Coma5	-0.012886	0.008832
// Sphere 5	0.011047	

History of measurements of FM-OTA WFE

date	0-G WFE (nm rms, single path)	Focus in 0G: A20 (lambda, double path)	Focus in 1G: A20 (lambda, double path)	Focus in -1G: A20 (lambda, double path)	test
2004.Jul.5-7	18.2	-1.29	-0.98	-1.63	FM-OTA initial integration
2004.Jul.14-17	d(5.2); hot mode –reference	/	/	-1.60(pre-test) -1.25(post-test)	Opto-thermal test III OTA downward
2004.Aug.	Note: WFE and focus (Zernike A20 term) given in the table are values measured at field center. WFE for OTA upward or downward configuration is given by a difference from a reference WFE obtained before (d()).				Sunlight test I
2004.Sep-Dec					System EIC/MIC test
2005.Jan.12-14	d (7.9); change by replacement	/	-1.78	/	Replacement of astig. Corrector OTA upward
2005.Mar.11-20	d(14.7); cold obs. mode – reference	/	/	-2.27(pre-test) -1.43(post-test)	Opto-thermal test IV with thermal cycling test OTA downward
2005.Apr					Acoustic test
2005.Apr.21-25	18.1	-1.74	-1.43	-2.06	
2005.Jun					Sunlight test II
2005.Sep.23	d (7.3); change from Apr.	/	-2.02	/	System Performance Test OTA upward

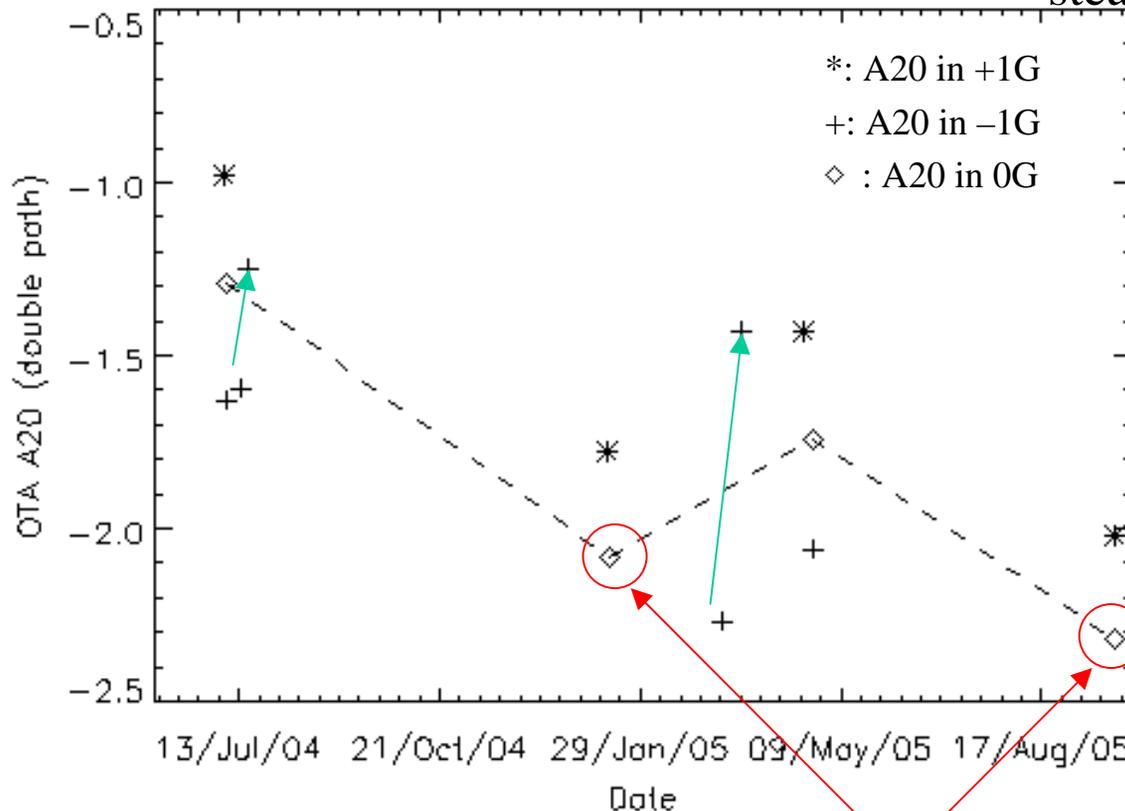
CFRP truss dehydration Focus Change

Truss contraction due to dehydration in vacuum.

Truss expansion due to moisture absorption in air.

A month after launch, the truss is expected to reach steady state

Focus change



↑ Truss contraction
↓ Truss expansion

Estimates from +1G measurement

Summary

OTA WFE at a room temperature (20 degC) for **zero-gravity** condition was **18 nm rms** when removed tilt and focus term and it did not change after a acoustic test and a thermal cycling test in which OTA experienced the coldest and hottest thermal environment on orbit; the **diffraction limited** WFE at wavelength of 500 nm is less than **35.7 nm rms**.

Change in WFE from the room temperature in vacuum to temperature on orbit for cold case observational condition was 14.5 nm rms (BOL) and this indicates total WFE on orbit is **23.1 nm rms**, combined with WFE of 18 nm rms for zero-gravity and room temperature condition.

The difference in WFE between **130 arcsec off-center** and the field center was 13.7 nm rms and hence WFE within 130 arcsec field radius is estimated to be less than **26.9 nm rms** including the thermal effect; (max FOV: 160×320 arcsec²).

Focus change shown here is due to vapor absorption (in air) or dehydration (in vacuum) effect of CFRP truss and stays in a range corrected on orbit by re-imaging lens of FPP.

OTA has a perfect optics!